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Swedish Radiation Safety Authority

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Research

2017:05

Robust structural verification of pressurized nuclear components subjected to ratcheting, part two – stress-strain curves for cyclic elastoplastic analysis.



## **SSM perspective**

### **Background**

Pressurized components in nuclear applications that are subjected to cyclic loading may exhibit progressive deformation, so called structural ratcheting. If the component is made out of a material that are deformation hardening, it may also exhibit material ratcheting.

The combined effects of structural- and material ratcheting are not taken into account in the methods and material models currently used for structural verification of pressurized nuclear components.

### **Objective**

The project aims to develop guidelines on how to evaluate pressurized nuclear components subjected to ratcheting in a rational and conservative way that are code compliant with ASME III.

In part one (SSM 2015:43) experimental studies where performed with two different laboratory test set ups, one called “two-rod test” and the other a tube test. Numerical studies with five different constitutive models where investigated for the ferritic steels P235 and P265 as well as the austenitic steel 316L. The constants in the constitutive models are based on material characterization via tensile testing and fully-reversed strain controlled cycling.

This report constitutes the second and final part of the project. It contains the experimental results of uniaxial testing of ferritic pressure vessel steels at elevated temperatures. It also contains stress-strain curves for stainless steels. These results give complementary information on material behaviour so that cyclic elastoplastic analysis, as specified in part one, can be performed to analyse ratcheting

### **Results**

- The presence of the yield plateau in the stress-strain curve of ferritic steels presents a problem when analysing ratcheting which is resolved with these results.
- Stress-strain curves for cyclic elastoplastic analysis of the ferritic pressure vessel steels 16M03 and P250GH are determined experimentally.
- The curves of 16M03 and P250GH are scaled to fit code values of strength according to SS-EN 10028.
- Information is also given for the ferritic pressure vessel steels P235GH, P265GH and P295GH as well as for the stainless steels 316L/1.4404 and 304L/1.4307.
- The curve data are also given in tabular form to allow for convenient input in finite element analysis software

### **Project information**

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Reference: SSM2015-3854





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This report concerns a study which has been conducted for the Swedish Radiation Safety Authority, SSM. The conclusions and viewpoints presented in the report are those of the author/authors and do not necessarily coincide with those of the SSM.

# SUMMARY

Herein, experimental results on uniaxial testing of ferritic pressure vessels steels at elevated temperatures are reported. The objective of these tests is to provide complementary information on material behaviour for the application of the results reached in the previously conducted research project ROBUS. More specifically, the presence of the yield plateau in the stress-strain curve of ferritic steels presents a problem for cyclic elasto-plastic analysis and this problem is resolved herein.

Stress-strain curves applicable for cyclic elastoplastic analysis are experimentally determined for 16Mo3 and P250GH. The curves are scaled so as to compensate for the over-strength observed in the experiments in comparison to code values of strength. Hence, the given stress-strain curves may be used directly for analysis. The curves are given also in tabular form so as to allow for a convenient input in finite element analysis software.

Information is given also for the steels P235GH, P265GH and P295GH. These steels were not tested due to inabilities of the workshop to manufacture circular specimen out of plates (oven test machine required circular specimen) and these steels cannot be delivered as round bars. However, they are similar to P250GH and generation of stress-strain curves via scaling of the ones for P250GH should be acceptable and this is done herein. All specimens for each material are manufactured from the same round bar and hence the statistical variation in properties between specimens is negligible. Therefore one specimen per curve is sufficient.

For the sake of completeness, stress-strain curves are given also for stainless steels 316L/1.4404 and 304L/1.4307. These steels do not exhibit the problematic yield plateau but pressure vessel codes in general do not give any information on stress-strain curves and hence such information is valuable enough to repeat herein. The information given herein for these steels is collected from the French nuclear code RCC-MRx, and is given in the same tabular form as the ferritic steels.

# SAMMANFATTNING

Föreliggande rapport redovisar enaxliga provningar av ferritiska tryckkärlsstål vid förhöjda temperaturer. Avsikten med dessa provningar är att generera kompletterande materialdata för tillämpning av de resultat som erhöles i projekt ROBUS. Mer specifikt är det förekomsten av flytplatå i arbetskurvan (spännings-töjningskurvan) hos ferritiska stål som är problematisk för cyklisk elastoplastisk analys och detta problem adresseras i denna rapport.

Arbetskurvor tillämpliga för cyklisk elastoplastisk analys har bestämts experimentellt för 16Mo3 och P250GH. Arbetskurvorna är skalade så att effekten av överstyrka i förhållande till normvärden på styrka är kompenserad för. De kan sålunda användas direkt i beräkning. Informationen är given även i tabellform vilket underlättar indata till beräkningsprogram.

Information är given också för P235GH, P265GH samt P295GH. Dessa stål är inte provade eftersom verkstaden inte kunde tillverka cirkulära provkroppar ur plåtar (provningssmaskinen i ugn fordrade cirkulära provkroppar) och dessa stål kan inte fås som rundstång. Dessa stål är dock lika P250GH och en skalning av resultaten för P250GH bör vara acceptabelt och detta är gjort här. Alla provkroppar för respektive material är svarvade ur samma stång och spridning av egenskaper mellan provkroppar är därför försumbar och ett prov per kurva är tillräckligt.

För fullständighetens skull redovisas även kurvor för rostfria stål 316L/1.4404 och 304L/1.4307. Dessa stål uppvisar inte den problematiska flytplatån men arbetskurvor anges i allmänhet inte i normer och sådan information är därmed värdefull och lämnas därför häri. De arbetskurvor som redovisas här är hämtade från den franska kärnkraftsnormen RCC-MRx.



# CONTENTS

<b>Summary</b> .....	<b>1</b>
<b>Sammanfattning</b> .....	<b>2</b>
<b>Contents</b> .....	<b>3</b>
<b>1. Introduction</b> .....	<b>4</b>
<b>2. Background</b> .....	<b>4</b>
<b>3. Experimental investigations</b> .....	<b>7</b>
3.1. Experimental setup .....	7
3.2. Monotonic tensile tests.....	8
3.3. Tensile to plateau extinct followed by load reversal .....	10
3.4. Stress-strain curves for cyclic elastoplastic analysis .....	16
<b>References</b> .....	<b>21</b>
<b>Appendix 1 Stress-strain curves for 316L / 1.4404</b> .....	<b>22</b>
<b>Appendix 2 Stress-strain curves for 304L / 1.4307</b> .....	<b>26</b>
<b>Appendix 3 Stress-strain curves for P265GH and P295GH</b> .....	<b>28</b>

# 1. Introduction

Herein, experimental results on uniaxial testing of ferritic pressure vessels steels are reported. The objective of these tests is to provide complementary information on material behaviour for the application of the results reached in the previously conducted research project ROBUS.

# 2. Background

Progressive deformation is one of three failure modes to address in the design and analysis of nuclear power plant pressure vessels and piping. Alternative denotations are ratcheting or incremental collapse. The remaining two failure modes are plastic collapse and low-cycle fatigue. All three involve plastic deformations. This suggests elastoplastic analysis would yield superior precision as compared to elastic analysis and that is indeed true for plastic collapse and low-cycle fatigue. The former is more or less independent of what constitutive model is adopted. The latter needs a kinematic hardening constitutive model but the type of kinematic model is more or less insignificant.

For progressive deformation however, there has been quite some scatter in numerical simulation reported in the literature. This is in distinct contrast to simulation of plastic collapse and low-cycle fatigue. Obviously, since progressive deformation is a case of cyclic loading in the plastic range, simulation requires constitutive models with kinematic elements. Different constitutive models have however been reported to predict very different results. This is indeed troublesome since the main objective with elastoplastic analysis is to provide accuracy.

There is indeed no lack of attempts to remedy this problem. In the literature there is a vast amount of proposals in terms of modification of constitutive models to better fit experimental data. In general, these modifications involve an increasing amount of model parameters. Constitutive models using 20 parameters or more have been presented. Although parameter adjustment of such models may result in agreement with a certain set of experimental data, the parameter setup is likely unsuitable for other loadings.

Such refined constitutive models obviously have an academic value. For industrial purposes however, the use of such models is not feasible. Constitutive models for industrial applications should be reasonably simple and involving as few model parameters as possible. For that purpose, the research project ROBUS (Robust structural verification of pressurized nuclear components subjected to ratcheting) was initiated by Areva NP Uddcomb and Inspecta. The aim of the project was to reach a conclusion on what constitutive model is feasible for ratchet simulation. The project was funded by SSM, RAB, OKG, FKA and TVO and is reported in [1].

In the project, extensive experimental investigations were conducted. The project was organized such that Inspecta conducted experiments and simulations of a Bree-type problem, and Areva NP Uddcomb conducted experiments and simulations of a pressurized

pipe subjected to axial strain cycling, corresponding to a pressurized pipe subjected to cyclic secondary bending. These subprojects were conducted independently of each other.

Simulations were conducted with the kinematic models of Prager, Armstrong-Frederick, Chaboche and Besseling. Moreover, simulations were conducted for ideal-elastoplastic material for comparison.

Armstrong-Frederick is a development of Prager and Chaboche is a development of Armstrong-Frederick. Subsequent more advanced models in general exhibit the same mathematical structure as Chaboche. Prager, Armstrong-Frederick and Chaboche are conventional kinematic models with a translating yield surface. The mathematical structure of Besseling is different. The material is assumed to be a composition of several ideal-elastoplastic subvolumes with different yield strengths. Hence, it does not involve yield surface translation. The kinematic feature follows from the build-up of internal residual stresses. This in fact resembles the real physical behaviour of metals on a crystal level.

In [1], a firm conclusion was reached. Simulations with the Besseling model were able to predict ratchet response with a surprisingly high accuracy. This was the case for both the Inspecta part of the project and for the Areva NP Uddcomb part. The number of different setups in the experimental tests was large and hence the agreement is not due to circumstance. The predictions from the Besseling model were far better than the predictions from the other models.

In addition to being more precise, there is a major benefit of the Besseling model. Whereas the others need adjustments of model parameters, the only input to the Besseling model is the stress-strain curve. The stress-strain curve is well defined and it needs no adjustments. As a result, there is only one possible outcome from a ratchet simulation.

As for the applicable stress-strain curve in ratchet simulation, the question arises whether to use the virgin monotonic stress-strain curve or the stabilized cyclic stress-strain curve. The latter is defined by subjecting a number of specimens to strain cycling at various strain amplitudes until the cyclic hardening is stabilized. The corresponding strain and stabilized stress amplitude constitutes a point in the stabilized cyclic stress-strain curve.

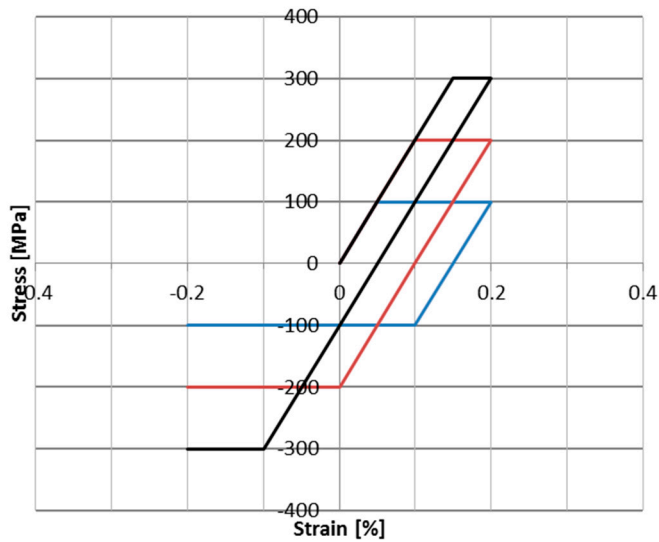
The cyclic curve is considerably more beneficial than the virgin monotonic one due to the cyclic hardening. However, it requires a number of cycles to reach the stabilized stress amplitude for each strain amplitude. Simulating a ratchet response of a virgin material using the virgin stress-strain curve will over-estimate the accumulated plastic strains. Conducting the same simulation with the stabilized cyclic stress-strain curve will under-estimate the accumulated plastic strains for a finite number of cycles. Because of this, the virgin stress-strain curve have been used in Sweden for ratchet simulations.

The virgin monotonic stress-strain curves of ferritic steels are however problematic in this respect. They exhibit a yield plateau which is unlike any other strain hardening metal. The yield plateau is a consequence of a complex interaction between interstitial and substitutional atoms around dislocations and really represents an artificial raise of the yield strength. Without the plateau, the stress-strain curve would be similar in shape to any

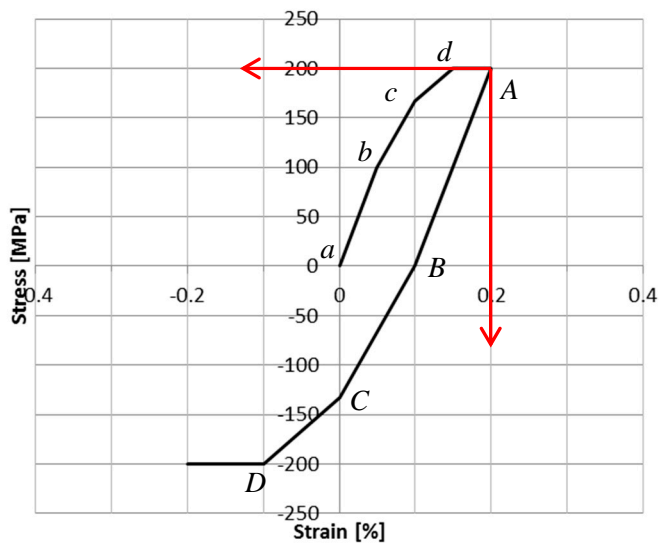
strain-hardening metal with considerably smaller yield strength  $R_{p0.2}$  than the actual yield strength  $R_{eL}$ .

Once the yield plateau is wiped out, the steel behaves as the plateau never existed. Hence, at a load reversal following tension to the point of yield plateau extinct, the reversed stress-strain curve is more or less identical in shape to the reversed stress-strain curve of stainless steel subjected to the same loading.

Obviously then, the applicable stress-strain curve of ferritic steels for simulation of cyclic elastoplastic loading is not the standard monotonic stress-strain curve. The applicable stress-strain curve may however be obtained as outlined already above. A uniaxial specimen is i) loaded in tension to the point of yield plateau extinct, whereafter ii) the loading is reversed. The stress-strain curve is taken as the unloading curve from the point of yield plateau extinct by which the stresses and strains are divided by 2 in accordance with the Masing's rule. Such curve was used in [1] for ratchet simulation with excellent results and the procedure is entirely compatible with the Besseling model as seen from the simplified illustration below. Assume a Besseling material with three equal subvolumes of elastic perfectly plastic (epp) materials according to Figure 1. The response in loading and subsequent load reversal is then according to Figure 2. Imagine stress-strain axes for the load reversal curve as indicated in Figure 2. Due to the build-up of internal residual stresses the load reversal curve  $ABCD$  comes out twice the loading curve  $abcd$ . Hence, the stress-strain curve  $abcd$  is reproduced taking the load reversal curve  $ABCD$  and dividing stresses and strains by 2.



**Figure 1.** Three epp materials with 100, 200 & 300 MPa yield strength.



**Figure 2.** Besseling response in loading and load reversal for three equally large subvolumes of materials according to Figure 1. Red stress-strain axes for the load reversal curve.

The objective of this work is to generate such curves at room temperature as well as elevated temperatures for steel 16Mo3 and P250GH. The latter grade was selected on the basis that it is the only PXGH steel available in round bar and it should represent P235GH – P295GH well enough, compensating for level of strength.

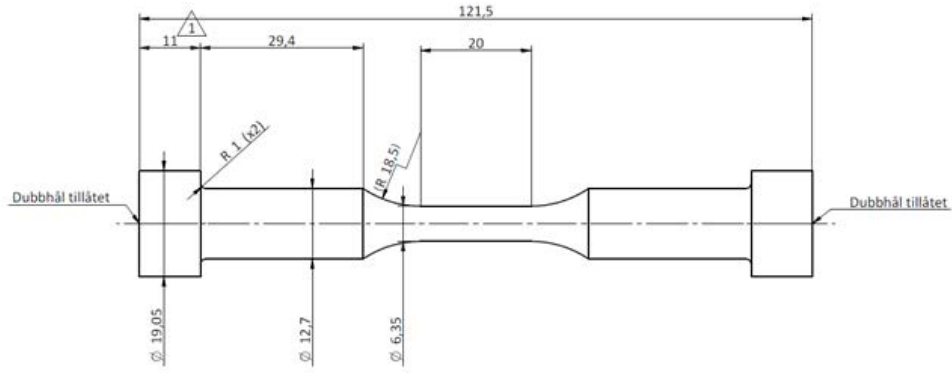
## 3. Experimental investigations

### 3.1. Experimental setup

The tensile tests were conducted at Lund University of technology, division of solid mechanics. The test specimen dimensions were according to Figure 3 below. The length 20 mm of the portion with diameter 6.35 mm was chosen as a compromise between obtaining an undisturbed stress field over the gauge length 12 mm and obtaining a buckling resistance enough to generate reasonably large compressive stresses and strains.

The tests at elevated temperatures were conducted in an oven as seen in Figure 4. Thermoelements was mounted so as to determine saturation of temperature.

The strain rate was determined so as to reach yield strength in approximately 30 seconds.



**Figure 3.** Test specimen.



**Figure 4.** Test setup in oven. Thermoelements to the left, displacement gauges to the right.

### 3.2. Monotonic tensile tests

The monotonic stress-strain curves obtained from tensile tests within this work are shown in Figure 5 for 16Mo3 and in Figure 6 for P250GH. These curves certainly exhibit some interesting and unexpected characteristics.

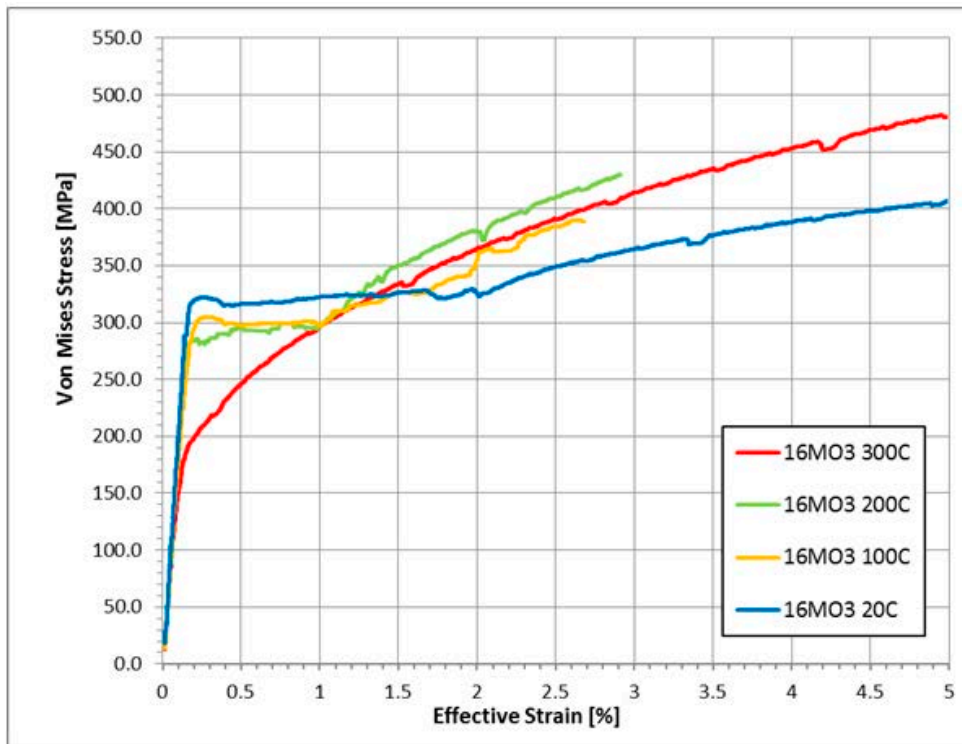
The first observation is that the yield strength decreases with increased temperature, as expected.

The second observation is that the extent of the yield plateau decreases with increased temperature. This was not expected but it is not really surprising since increased temperature increases dislocation mobility.

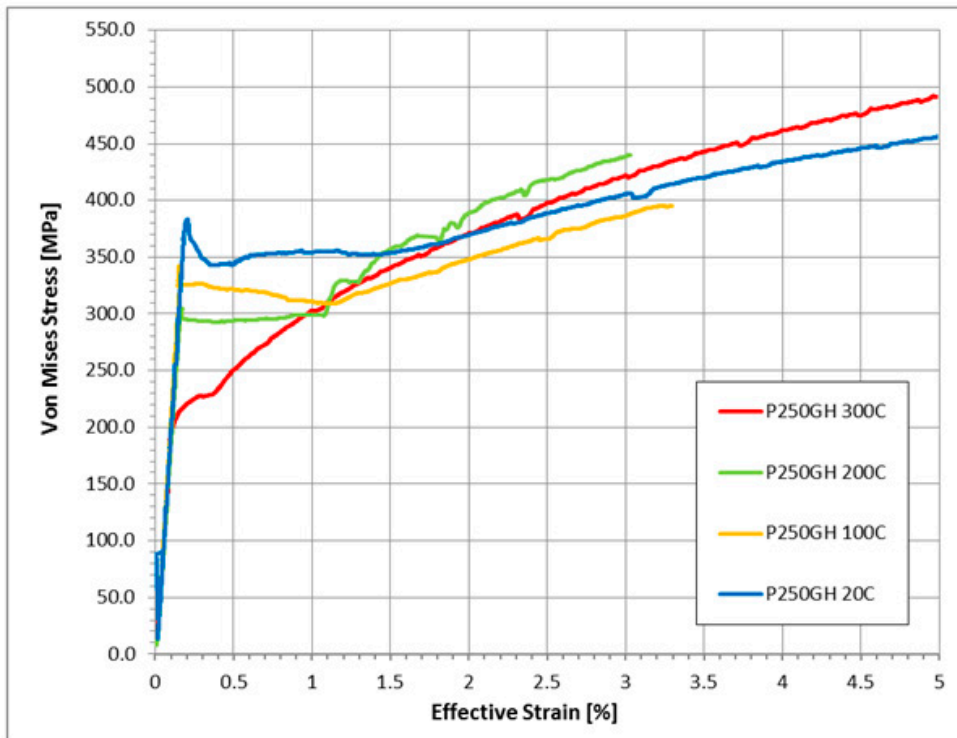
Third observation is that for strains beyond the yield plateaus, in the < 5 % range, the stress is higher for 300 °C and 200 °C than for 20 °C for P265GH. For 16Mo3, also the stress for 100 °C is higher than for 20 °C. This was certainly unexpected. In fact, it was so unexpected that initially it was believed that the tests were erroneously conducted.

The French code for experimental and high-temperature reactors RCC-MRx, [2] , is an invaluable source of material data. It contains monotonic stress-strain curves for P265GH and for P295GH. These are shown in Figure 22 and Figure 23, respectively, in Appendix 3. As seen, in the < 5 % range, the same behavior is reported in the RCC-MRx for P265GH and P295GH as the observed behavior herein for P250GH. This behavior is particularly clear for P295GH.

The objective of these monotonic stress-strain curves was to determine the strain magnitudes at stress reversal for the subsequent tests. For instance, the length of the 16Mo3 yield plateau for 20 °C is seen to be 2 % in Figure 5 below and hence stress reversal (red curve) in Figure 7 is conducted at 2 %.



**Figure 5.** Stress-strain curves at room temperature and elevated temperatures for 16Mo3 as obtained from tensile tests within this work.



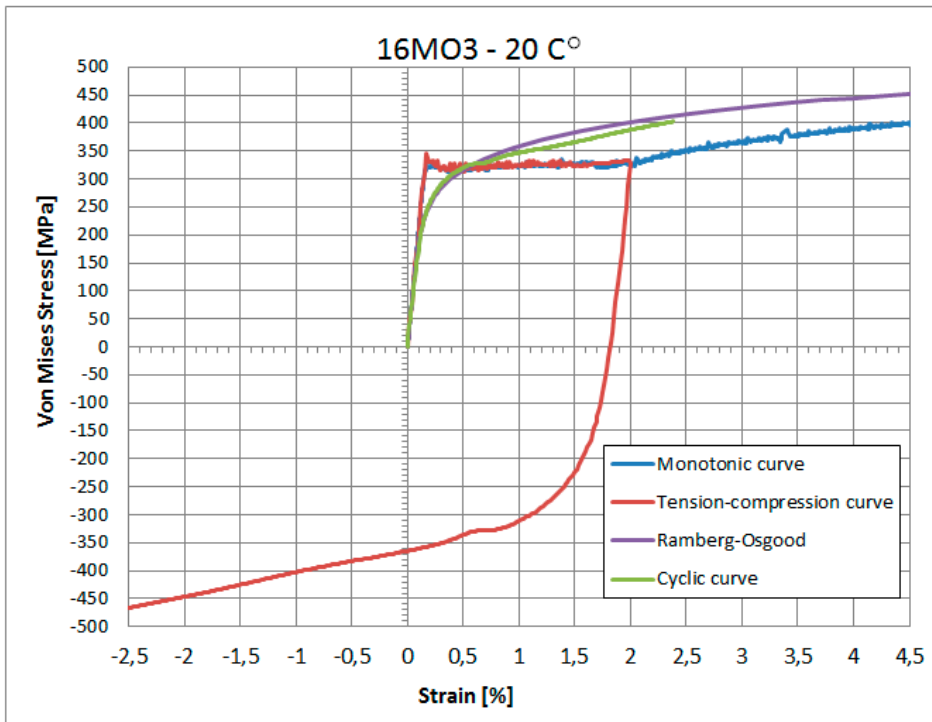
**Figure 6.** Stress-strain curves at room temperature and elevated temperatures for P250GH as obtained from tensile tests within this work.

### 3.3. Tensile to plateau extinct followed by load reversal

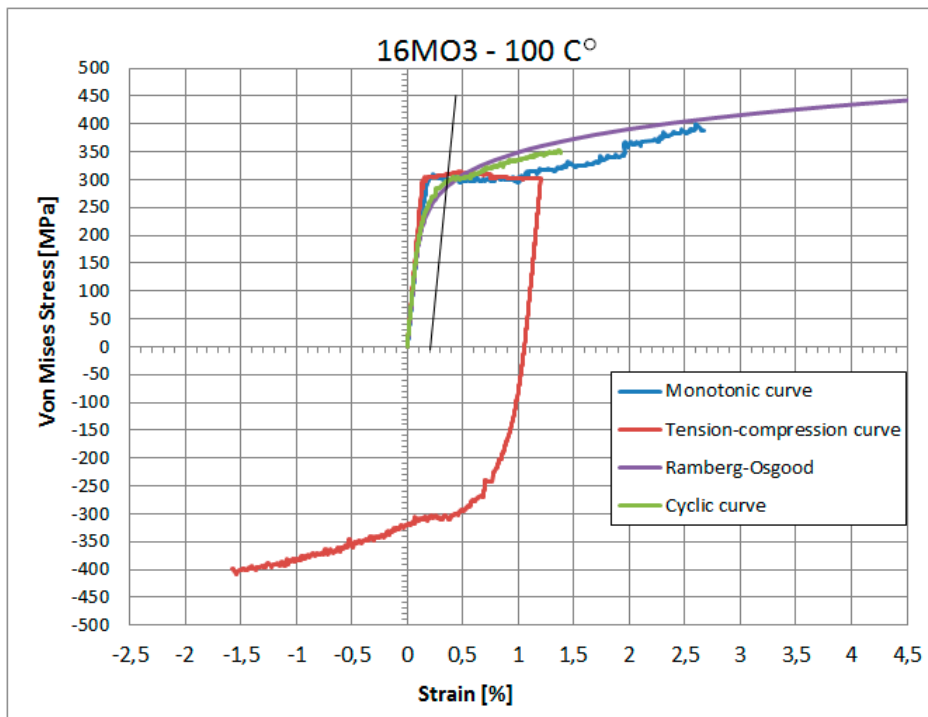
The tension to extinct of yield plateau followed by load reversal tests are conducted analogously to the monotonic tests. The results for 16Mo3 are shown in Figure 7 – Figure 10. In these figures the monotonic tests are included as well as the unloading curve divided by 2 as discussed above (denoted in the figures as cyclic curve), and a Ramberg-Osgood fit of this unloading curve. The corresponding graphs for P250GH are shown in Figure 11 – Figure 14. In Figure 15 and Figure 16 the unloading curves divided by 2 (denoted in the figures as cyclic curve) are given for 16Mo3 and P250GH, respectively.

These unloading curves constitute the applicable stress-strain curves to be used for cyclic elastoplastic analysis. Their magnitudes however need to be corrected so as correspond to code strength values. This is conducted in the following section.

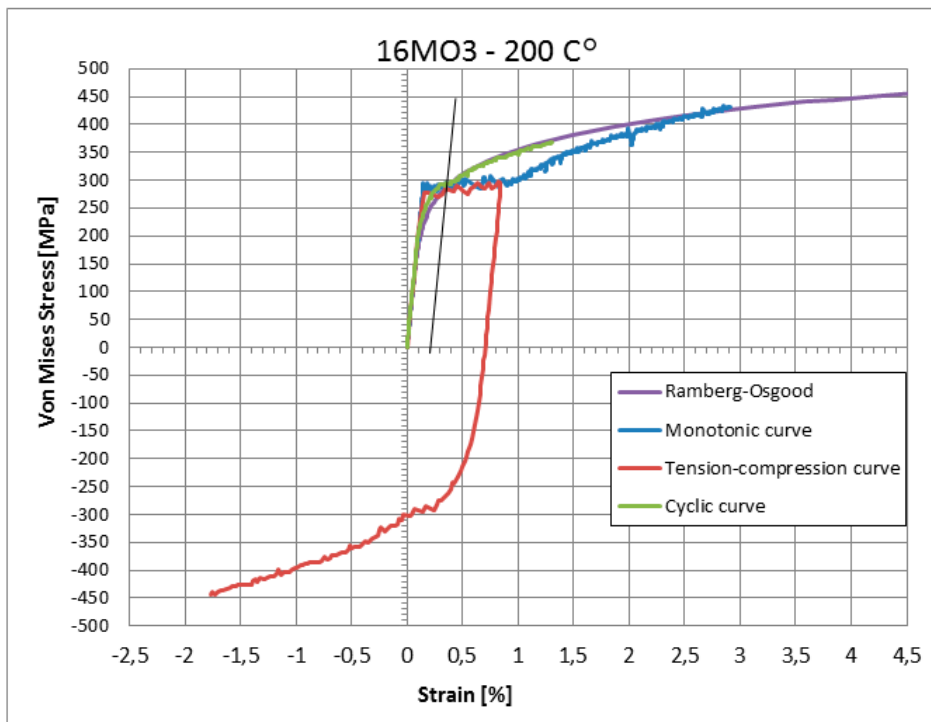




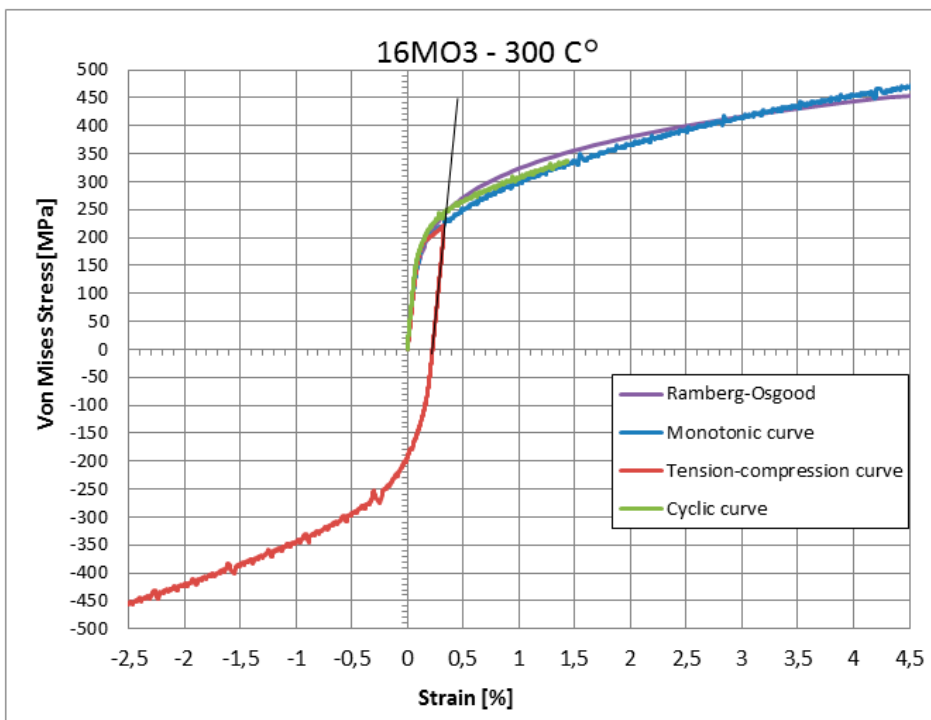
**Figure 7.** 16Mo3 at 20 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).



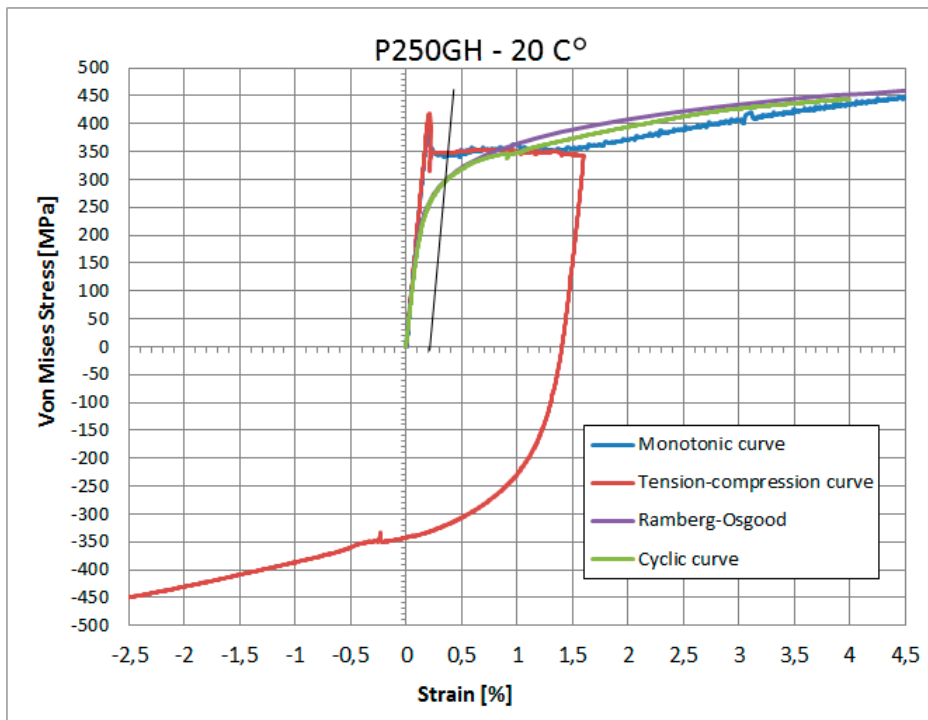
**Figure 8.** 16Mo3 at 100 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple). Thin black line indicates 0.2 %.



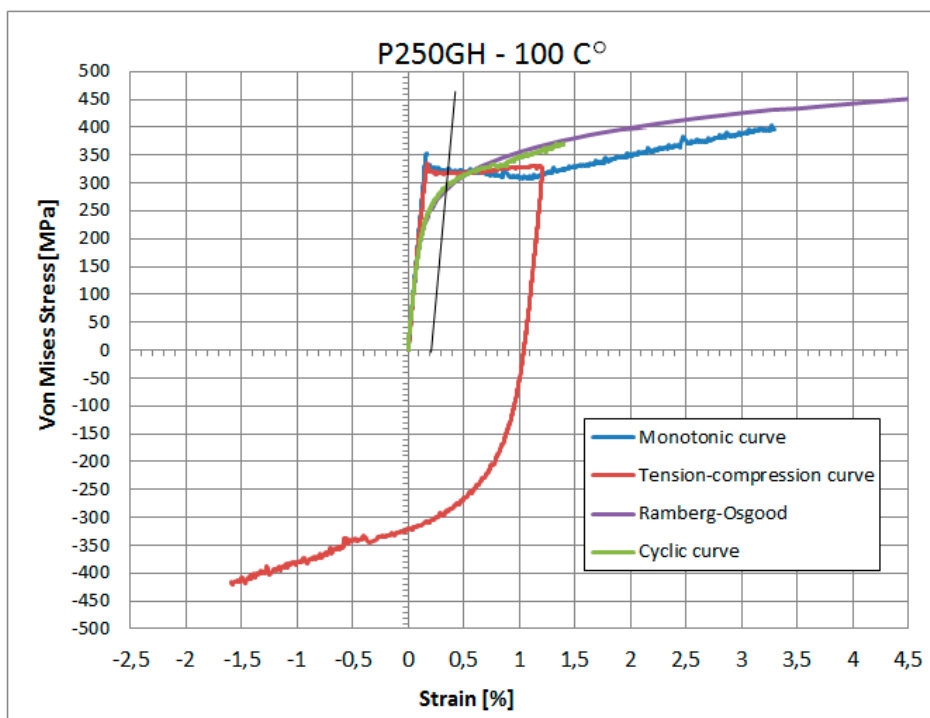
**Figure 9.** 16Mo3 at 200 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).



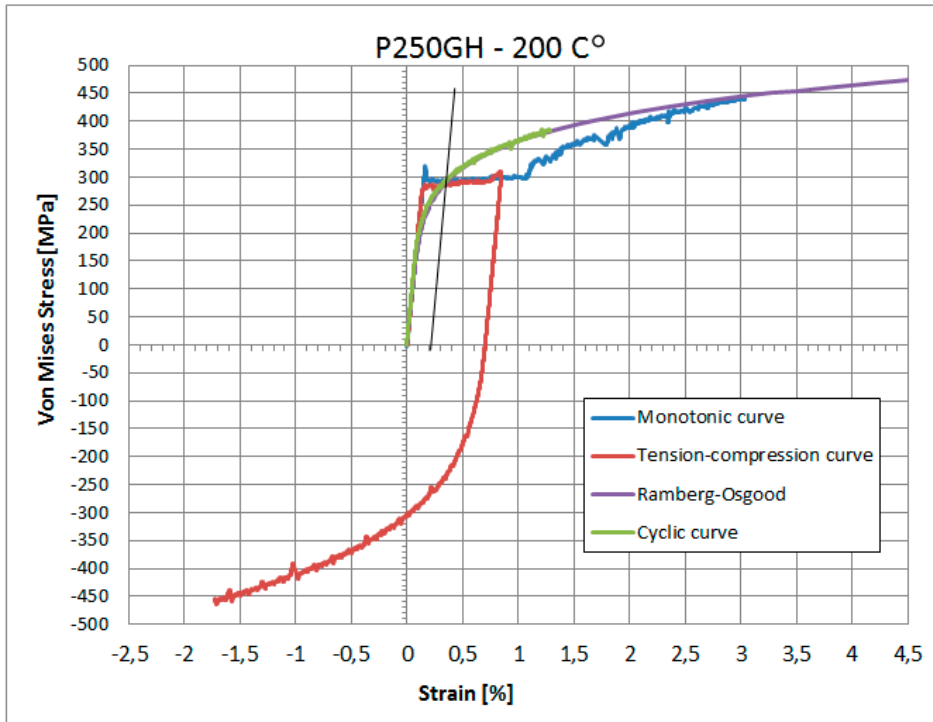
**Figure 10.** 16Mo3 at 300 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).



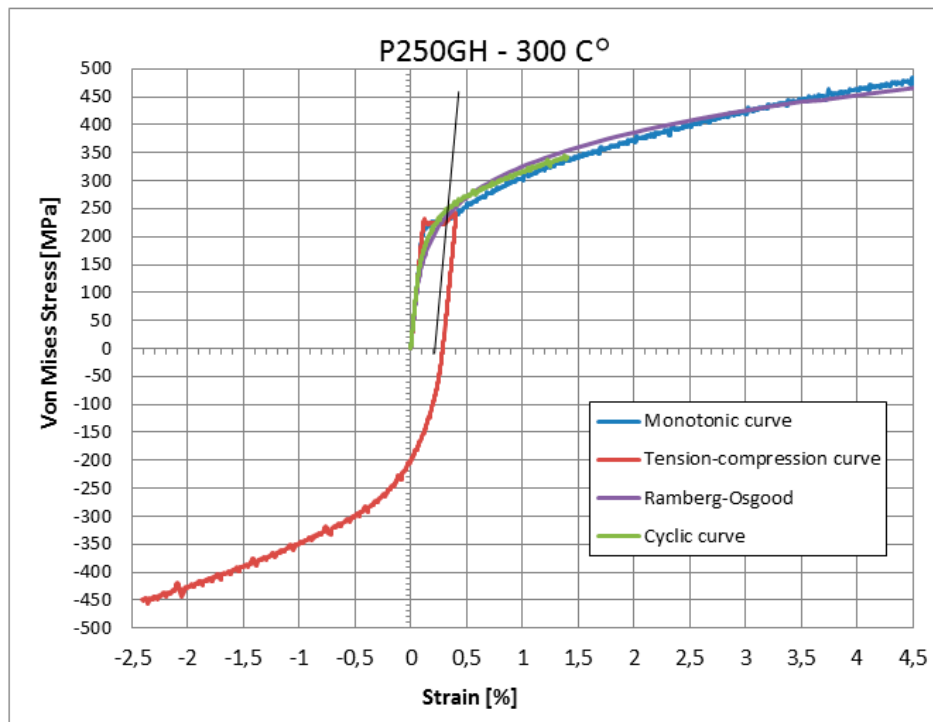
**Figure 11.** P250GH at 20 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).



**Figure 12.** P250GH at 100 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).



**Figure 13.** P250GH at 200 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).



**Figure 14.** P250GH at 300 °C. Monotonic curve (blue), loading plus unloading curve (red), loading plus unloading divided by two and moved to origo (green), and Ramberg-Osgood fit to the green (purple).

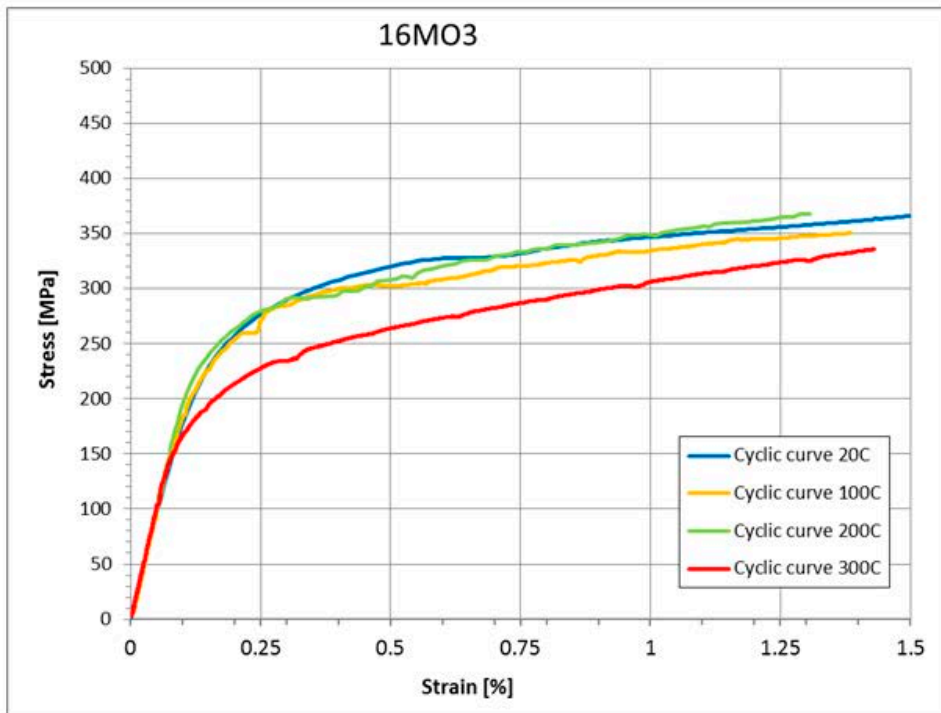


Figure 15. 16Mo3. Reversed stress-strain curve divided by 2.

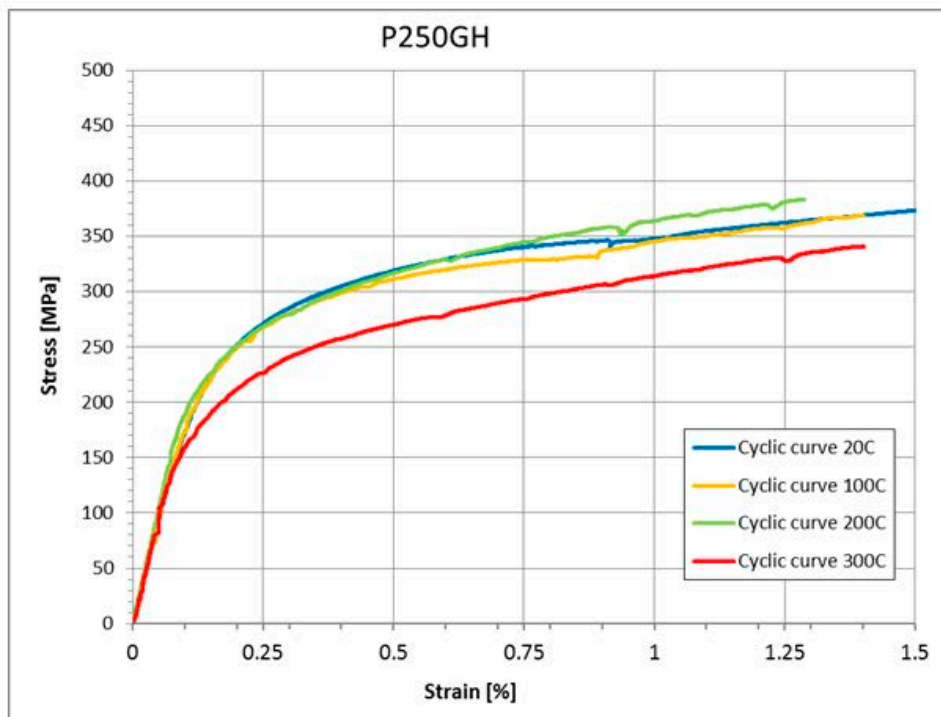


Figure 16. P250GH. Reversed stress-strain curve divided by 2.

### 3.4. Stress-strain curves for cyclic elastoplastic analysis

The stress-strain curves in Figure 15 and Figure 16 are applicable for cyclic elastoplastic analysis of 16Mo3 and P250GH components, respectively, but their magnitudes correspond to the particular batch they are made from and hence they do not represent code magnitudes of strength. As seen in Figure 5 and Figure 6, the experimental yield strengths at room temperature are  $S_y = 320$  MPa for 16Mo3 and  $S_y = 350$  MPa for P250GH. The corresponding code yield strength values are  $S_y = 275$  MPa for 16Mo3 and  $S_y = 250$  MPa for P250GH according to SS-EN 10028. Hence, to obtain room temperature stress-strain curves with strength level according to code values, the experimental stress-strain curves are scaled with the factors  $275 / 320$  and  $250 / 350$  for 16Mo3 and P250GH, respectively.

Now, as for the corresponding curves at elevated temperatures, there is no particular correspondence between the monotonic curves and the unloading curves. This appears to be due to the varying magnitudes for the yield plateau. Therefore, to obtain elevated temperatures stress-strain curves with code strength levels, these are scaled with the same factors as the room temperature curves.

The experimental investigation covers 16Mo3 and P250GH. As seen, there are only small differences between the two. The behavior of steels P235GH, P265GH and P295GH are likely to be very similar to P250GH and to obtain applicable stress-strain curves for these materials, the stress-strain curves for P250 GH are scaled with factors  $235 / 250$ ,  $265 / 250$  and  $295 / 250$ , respectively.

The stress-strain curves applicable for cyclic elastoplastic analysis are given in Figure 17 - Figure 21 below. The information is given also as tabular data in Table 1 – Table 6. It is recognized that the stress-strain curves for temperatures 20°C, 100°C and 200°C in Figure 17 – Figure 21 are more or less the same. The minor deviations are not worthwhile to account for and therefore one curve is used for temperatures up to 200°C in the tabular data.

The stress-strain curves in Figure 17 – Figure 21 go to 1.5 % strain. However, for the 20 °C case experimental data covers 5 %. The reason for the others not reaching 5 % was due to buckling of the specimen. The same tangent modulus is therefore assumed for elevated temperatures between 1.5 and 5 % strain, which allows for tabular data up to 5 % strain for all temperatures. Larger strain than that is not required for stress-strain curves used for cyclic elastoplastic analysis.

For completeness, this report gives information also about stress-strain curves applicable for stainless steels. The information given herein is collected from [2] and applies to steel X2CrNiMo17-12-2 which is a 1.4404 steel corresponding approximately to 316L, and X2CrNi18-9 which is a 1.4307 steel corresponding roughly to 304L.

The derivation of the tabular data for stress-strain curves for these steels are conducted in Appendix 1 and Appendix 2. The results are shown in Table 7 and Table 8 below. It should be mentioned the strength data is collected from [2] and the strength magnitudes are slightly low which may be due to the fact that there are no limitations on thicknesses. For applications where higher strength is applicable, the stress-strain curves may be scaled by the ratio of yield strengths. This is compatible with the rules given in [2] .

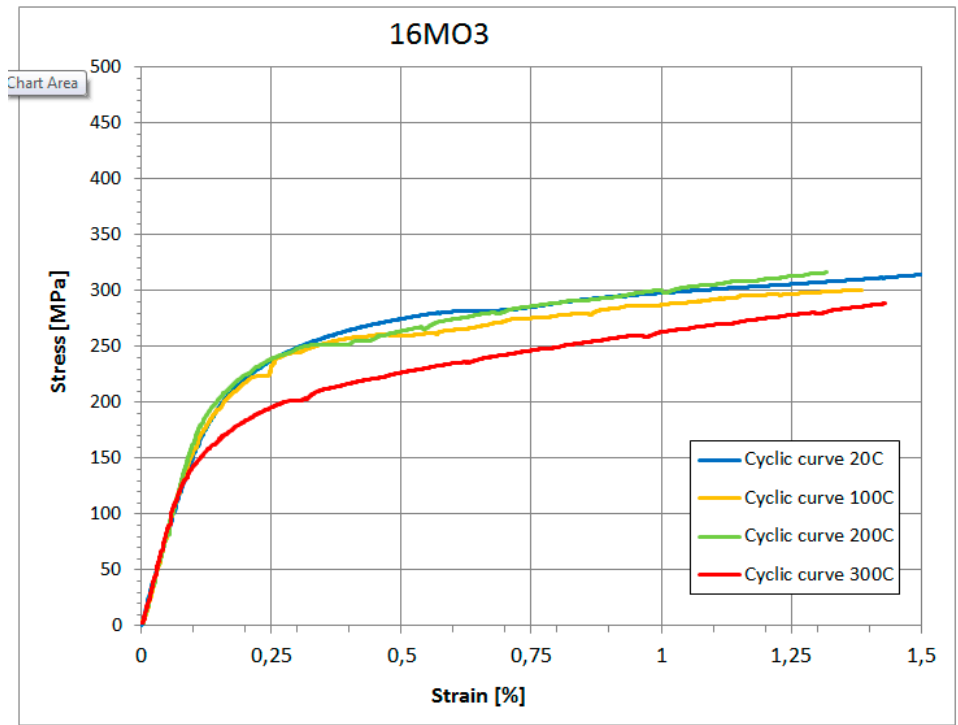


Figure 17. Stress-strain curves applicable for cyclic elastoplastic analysis of 16Mo3 components.

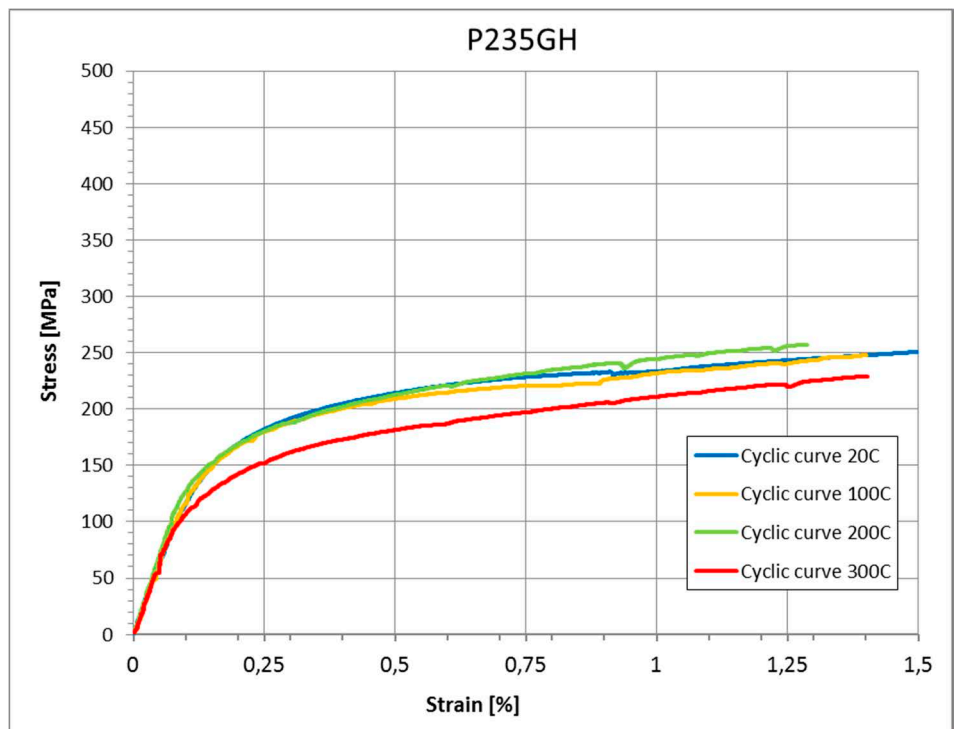
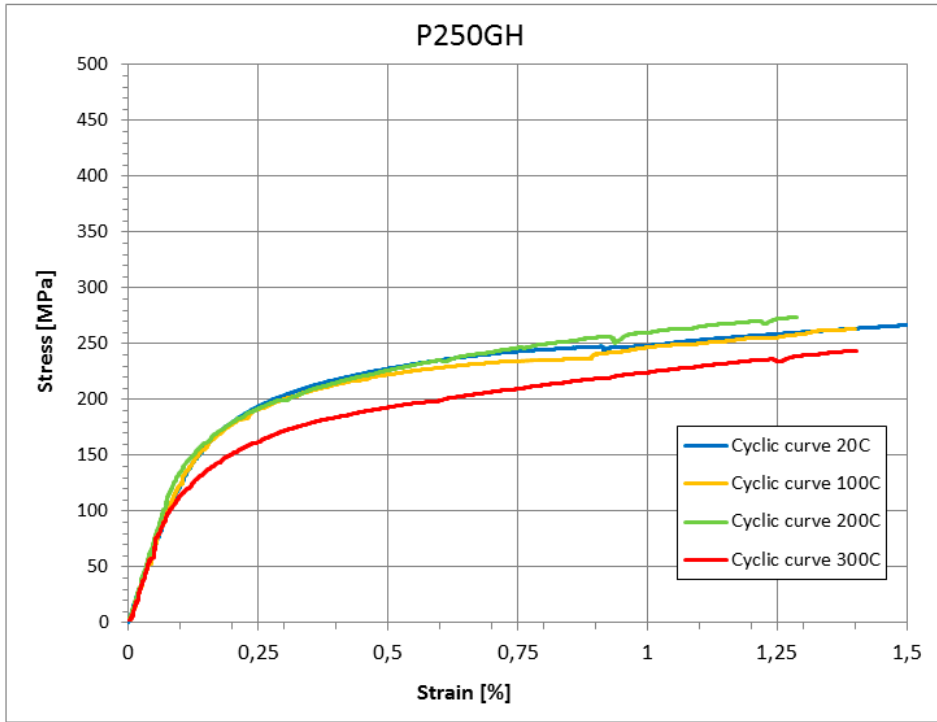
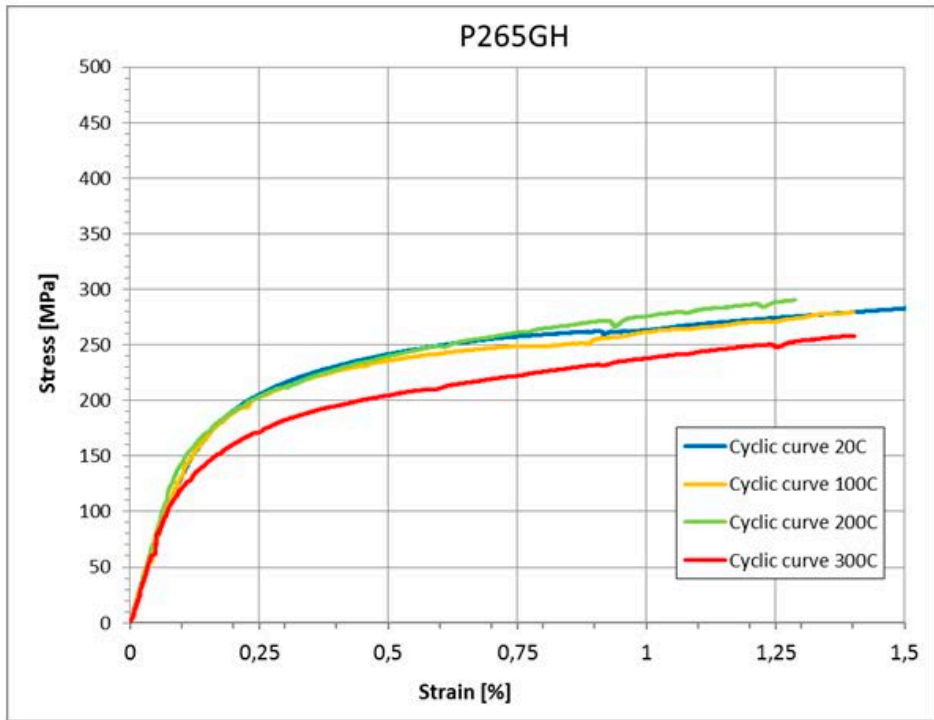


Figure 18. Stress-strain curves applicable for cyclic elastoplastic analysis of P235GH components.

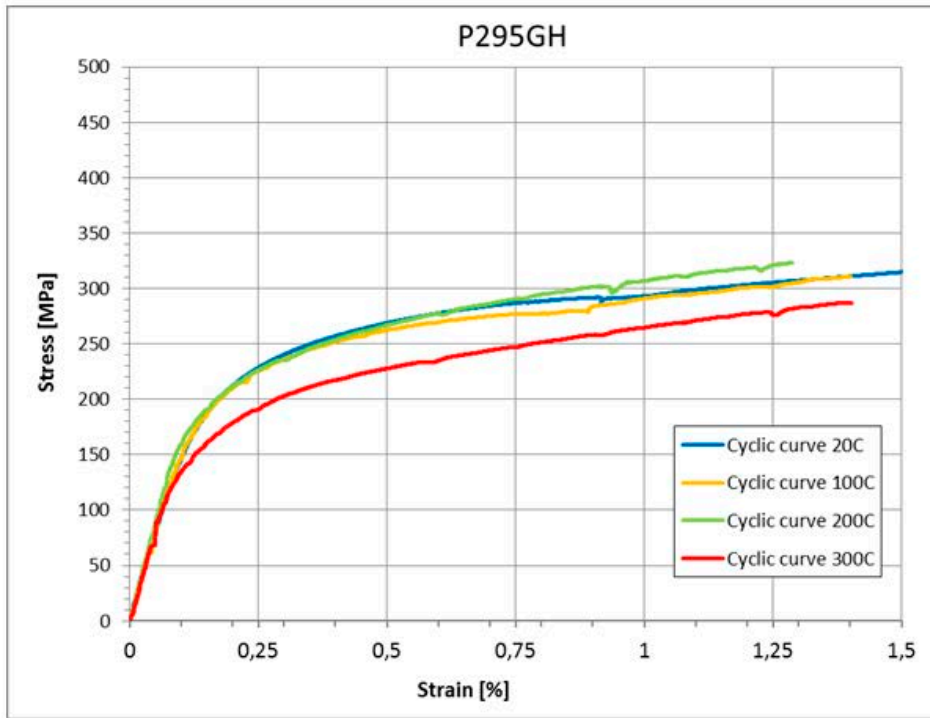


**Figure 19.** Stress-strain curves applicable for cyclic elastoplastic analysis of P250GH components.



**Figure 20.** Stress-strain curves applicable for cyclic elastoplastic analysis of P265GH components.





**Figure 21.** Stress-strain curves applicable for cyclic elastoplastic analysis of P295GH components.

**Table 1.** Tabular data point 1 for stress-strain curves applicable for cyclic elastoplastic analysis of all ferritic components.

$T$ [°C]	$\varepsilon$	$\sigma$
< 200	0.0005	100
300	0.0005	95

**Table 2.** Tabular data points 2 – 7 for stress-strain curves applicable for cyclic elastoplastic analysis of 16Mo3 components.

$\varepsilon$	0,001	0,002	0,003	0,005	0,01	0,05
$T$						
< 200 °C	160 MPa	225 MPa	250 MPa	270 MPa	300 MPa	370 MPa
300 °C	140 MPa	180 MPa	205 MPa	225 MPa	255 MPa	325 MPa

**Table 3.** Tabular data points 2 – 7 for stress-strain curves applicable for cyclic elastoplastic analysis of P235GH components.

$\varepsilon$	0,001	0,002	0,003	0,005	0,01	0,05
$T$						
< 200 °C	122 MPa	169 MPa	188 MPa	212 MPa	235 MPa	301 MPa
300 °C	108 MPa	146 MPa	160 MPa	179 MPa	207 MPa	273 MPa

**Table 4.** Tabular data points 2 – 7 for stress-strain curves applicable for cyclic elastoplastic analysis of P250GH components.

$T \backslash \varepsilon$	0,001	0,002	0,003	0,005	0,01	0,05
< 200 °C	130 MPa	180 MPa	200 MPa	225 MPa	250 MPa	320 MPa
300 °C	115 MPa	155 MPa	170 MPa	190 MPa	220 MPa	290 MPa

**Table 5.** Tabular data points 2 – 7 for stress-strain curves applicable for cyclic elastoplastic analysis of P265GH components.

$T \backslash \varepsilon$	0,001	0,002	0,003	0,005	0,01	0,05
< 200 °C	138 MPa	191 MPa	212 MPa	239 MPa	265 MPa	339 MPa
300 °C	122 MPa	164 MPa	180 MPa	201 MPa	233 MPa	307 MPa

**Table 6.** Tabular data points 2 – 7 for stress-strain curves applicable for cyclic elastoplastic analysis of P295GH components.

$T \backslash \varepsilon$	0,001	0,002	0,003	0,005	0,01	0,05
< 200 °C	153 MPa	212 MPa	236 MPa	266 MPa	295 MPa	378 MPa
300 °C	136 MPa	183 MPa	201 MPa	224 MPa	260 MPa	342 MPa

**Table 7.** Tabular data for stress-strain curves applicable for cyclic elastoplastic analysis of 316L / 1.4404 components.

$T \backslash \varepsilon$	0,00049	0,00054	0,00061	0,00070	0,002	0,003	0,005	0,01	0,05
20 °C	-	-	-	140 MPa	174 MPa	186 MPa	202 MPa	224 MPa	297 MPa
100 °C	-	-	119 MPa	-	153 MPa	164 MPa	177 MPa	194 MPa	275 MPa
200 °C	-	100 MPa	-	-	129 MPa	138 MPa	146 MPa	162 MPa	238 MPa
300 °C	86 MPa	-	-	-	112 MPa	118 MPa	129 MPa	140 MPa	212 MPa

**Table 8.** Tabular data for stress-strain curves applicable for cyclic elastoplastic analysis of 304L / 1.4307 components.

$T \backslash \varepsilon$	0,00042	0,00046	0,00054	0,00066	0,002	0,003	0,005	0,01	0,05
20 °C	-	-	-	131 MPa	163 MPa	175 MPa	190 MPa	210 MPa	280 MPa
100 °C	-	-	105 MPa	-	135 MPa	144 MPa	155 MPa	171 MPa	242 MPa
200 °C	-	86 MPa	-	-	111 MPa	118 MPa	126 MPa	139 MPa	204 MPa
300 °C	73 MPa	-	-	-	95 MPa	100 MPa	110 MPa	119 MPa	180 MPa

# REFERENCES

- [1] Möller, M., Gustafsson, A., Segle, P., *Robust structural verification of pressurized nuclear components subjected to ratcheting*. SSM research report 2015:43.
- [2] *RCC-MRx. Design and construction rules for mechanical components of nuclear installations*. Afcen, 2012 edition.

# APPENDIX 1 STRESS-STRAIN CURVES FOR 316L / 1.4404

Information on monotonic stress-strain curves for stainless steel 316L / 1.4404 may be collected from the French nuclear code RCC-MRx, [2] . The information given below is taken from Appendix 3 of that code. The section A3.3S.451 gives stress-strain curves up to 1 % and the section A3.3S.452 gives stress-strain curves up to ultimate strain, however up to 5 % strain is sufficient for elastoplastic cyclic analysis. Section A3.3S.452 gives stress-strain curves also in tabular data.

All given curves below are average curves. The average and minimum yield strengths  $R_{0.2,moy}$  and  $R_{0.2,min}$  are given in table A3.3S.41 below. As seen in section A3.3S.451 below, the minimum stress-strain curves are obtained simply by scaling the average ones by the factor  $R_{0.2,min} / R_{0.2,moy}$  . Doing so, the obtained applicable stress-strain curves for cyclic elastoplastic analysis are given in Table 9 below.

The information may be used also for 316 / 1.4401.

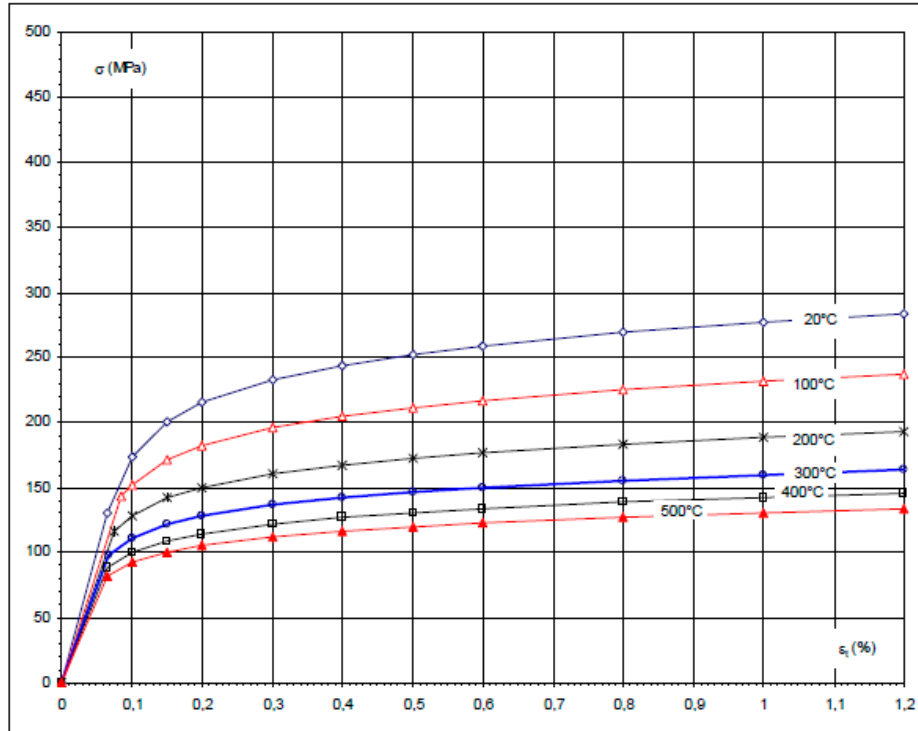
### A3.3S.451 FOR PLASTIC STRAIN LIMITED TO 1%

- The average tensile stress-strain curves are given by the formula A3.3S.451a and the figure A3.3S.451.
  - Formula A3.3S.451a :**  

$$\sigma = C_0 \times (R_{p0,2}^1)_{\text{avg}} \times (\epsilon_p)^{n_0} \quad \epsilon_t = 100 \times \sigma / E + [ \sigma / (C_0 \times (R_{p0,2}^1)_{\text{avg}}) ]^2$$
 In this formulae,  $\epsilon_p$  (%) and  $\epsilon_t$  (%) designate respectively the plastic strain and the total strain induced by stress  $\sigma$  (MPa).  $(R_{p0,2}^1)_{\text{avg}}$  is given in A3.3S.41.  
 The parameter  $n_0$  is equal to 0.1125 ;  $\alpha = 1/n_0$   
 The constant  $C_0$  is equal to 1.198.
- The minimum tensile stress-strain curves are given by the formula A3.3S.451b.
  - Formula A3.3S.451b :**  

$$\sigma = C_0 \times (R_{p0,2}^1)_{\text{min}} \times (\epsilon_p)^{n_0} \quad \epsilon_t = 100 \times \sigma / E + [ \sigma / (C_0 \times (R_{p0,2}^1)_{\text{min}}) ]^2$$
 with the same designations as for average tensile stress-strain curves.  
 $(R_{p0,2}^1)_{\text{min}}$  is given in A3.3S.41.  
 Same values of  $C_0$  and  $n_0$  as for average tensile stress-strain curves.

Figure A3.3S.451: average tensile stress-strain curves



### A3.3S.452 FOR TOTAL STRAIN ATTAINING MAXIMUM ELONGATION

The average rational tensile stress-strain curves at a given temperature are schematically shown by a polygonal contour OABCDE. The coordinates of points A, B, C, D and E are given as a function of temperature  $\theta$  by the table A3.3S.452 and the figure A3.3S.452:

- part OABC ( $\epsilon_t \leq 1\%$ ): average tensile stress-strain curve given in A3.3S.451.
- part CD ( $1\% \leq \epsilon_t \leq 5\%$ ): straight line  $\sigma_{1\%} - \sigma_{5\%}$  where  $\sigma_{1\%}$  and  $\sigma_{5\%}$  designate the stresses with respectively 1% and 5% of total strain,
- part DE ( $5\% < \epsilon_t \leq \epsilon_u$ ): straight line  $\sigma_{5\%} - \sigma_u$  where  $\sigma_u$  and  $\epsilon_u$  designate the true stress and the true strain for the maximum load.

Table A3.3S.452: average tensile rational stress-strain curves

$\theta(^{\circ}\text{C})$	A		B		C		D		E	
	$\epsilon_t$ (%)	$\sigma$ (MPa)	$\epsilon_t$ (%)	$\sigma$ (MPa)	$\epsilon_t$ (%)	$\sigma$ (MPa)	$\epsilon_t$ (%)	$\sigma$ (MPa)	$\epsilon_u$ (%)	$\sigma_u$ (MPa)
20	0.100	173.3	0.2	215.3	1	276.8	5	367.9	36.5	763
100	0.086	142.8	0.2	182.3	1	231.5	5	327	31.8	646
200	0.075	116	0.2	149.9	1	188.2	5	275.5	29.3	576
300	0.068	98.45	0.2	128.2	1	160	5	242	28.4	562
400	0.066	88.42	0.2	114.5	1	142.3	5	228.1	28	560
500	0.066	81.9	0.2	105.1	1	130.5	5	222.9	28	544

Figure A3.3S.452: average tensile rational stress-strain curves

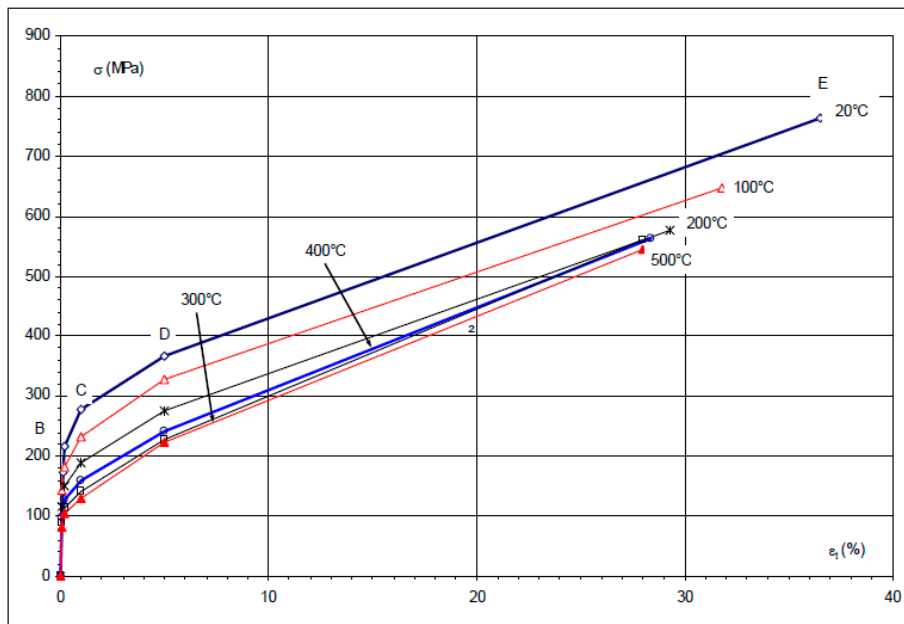


Table A3.3S.41: values of  $R_{p0.2}^i$ ,  $R_m$ ,  $S_m$  and  $S$

Temperature ( $^{\circ}\text{C}$ )	$R_{p0.2\text{min}}$ (MPa)	$R_{p0.2\text{moy}}$ (MPa)	$R_{m\text{min}}$ (MPa)	$R_{m\text{moy}}$ (MPa)	$S_m$ (MPa)	$S$ (MPa)
20	190	235	490	530	127	123
100	165	196	430	470	127	119
150	150				127	114
200	137	159	390	430	123	108
250	127				114	107
300	118	135	380	423	106	106
350	113	126	380	423	102	102
400	108	120	380	423	97	97
450	103	115	370	420	93	93
500	100	110	360	411	90	90
550	98	106	350	388	88	88

**Table 9.** Tabular data for stress-strain curves applicable for cyclic elastoplastic analysis of 316L / 1.4404 components. The information may be used also for 316 / 1.4401.

$T \backslash \varepsilon$	0,00049	0,00054	0,00061	0,00070	0,002	0,003	0,005	0,01	0,05
20 °C	-	-	-	140 MPa	174 MPa	186 MPa	202 MPa	224 MPa	297 MPa
100 °C	-	-	119 MPa	-	153 MPa	164 MPa	177 MPa	194 MPa	275 MPa
200 °C	-	100 MPa	-	-	129 MPa	138 MPa	146 MPa	162 MPa	238 MPa
300 °C	86 MPa	-	-	-	112 MPa	118 MPa	129 MPa	140 MPa	212 MPa

## APPENDIX 2 STRESS-STRAIN CURVES FOR 304L / 1.4307

Information on monotonic stress-strain curves for stainless steel 304L / 1.4307 may be collected from the French nuclear code RCC-MRx, [2] as for 316L / 1.4404 in Appendix 1 above. The section A3.4S.451 below gives stress-strain curves up to 1 %. The corresponding information on stress-strain curves for strains exceeding 1 % as given for 316L / 1.4404 in section A3.3S.452 is lacking for 304L / 1.4307.

However, noting that in the mathematical representation of stress-strain curves in sections A3.3S451 and A3.4S451, the only parameter that differs is the yield strength. Hence, the applicable stress-strain curves for cyclic elastoplastic analysis of 304 L / 1.4307 components are obtained simply by scaling the yield strengths. The yield strengths at temperatures for 304L / 1.4307 are given in table A3.4S.41 below.

The information may be used also for 304 / 1.4301.

### A3.4S.451 FOR PLASTIC STRAIN LIMITED TO 1%

- The average tensile stress-strain curves are given by the formula A3.4S.451a and the figure A3.4S.451.
  - Formula A3.4S.451a : 
$$\sigma = C_0 \times (R_{p0.2}^i)_{\text{moy}} \times (\varepsilon_p)^{\alpha} \quad \varepsilon_i = 100 \times \sigma / E + [ \sigma / (C_0 \times (R_{p0.2}^i)_{\text{moy}}) ]^2$$

In this formula,  $\varepsilon_p$  (%) and  $\varepsilon_i$  (%) designate respectively the plastic strain and the total strain induced by stress  $\sigma$  (MPa).  $(R_{p0.2}^i)_{\text{moy}}$  is given in A3.4S.41.  
The parameter  $n_0$  is equal to 0.1125 ;  $\alpha = 1/n_0$   
The constant  $C_0$  is equal to 1.198.
- The minimum tensile stress-strain curves are given by the formula A3.4S.451b.
  - Formula A3.4S.451b : 
$$\sigma = C_0 \times (R_{p0.2}^i)_{\text{min}} \times (\varepsilon_p)^{\alpha} \quad \varepsilon_i = 100 \times \sigma / E + [ \sigma / (C_0 \times (R_{p0.2}^i)_{\text{min}}) ]^2$$

with the same designations as for average tensile stress-strain curves.  
 $(R_{p0.2}^i)_{\text{min}}$  is given in A3.4S.41.  
Same values of  $C_0$  and  $n_0$  as for average tensile stress-strain curves.

Figure A3.4S.451: average tensile stress-strain curves

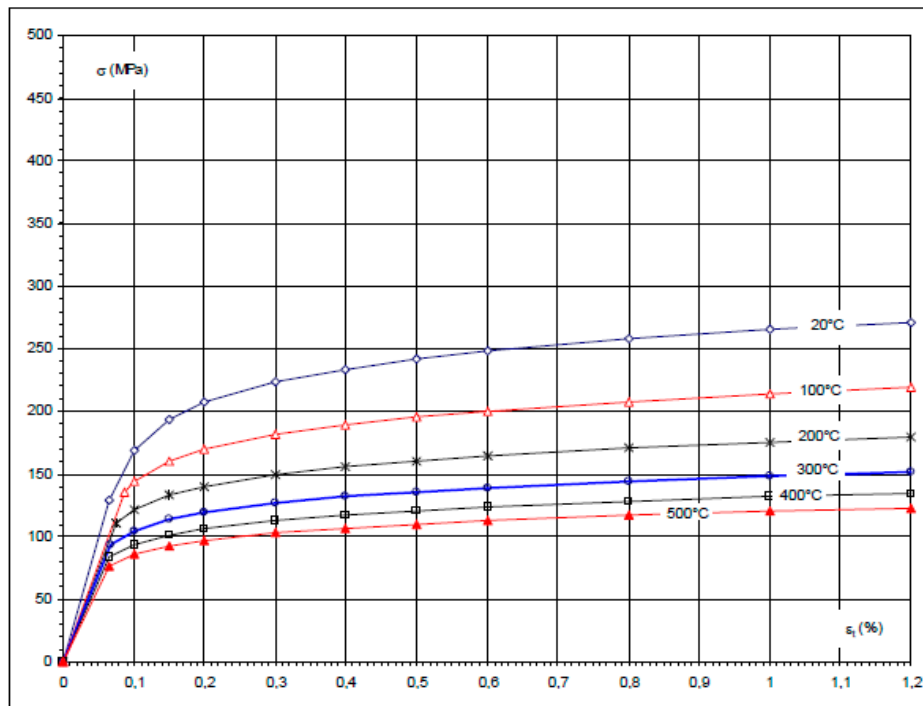




Table A3.4S.41: values of  $R_{p0,2}$ ,  $R_m$ ,  $S_m$  and  $S$

Temperature (°C)	Products and parts made from solution-annealed X2CrNi18-9 and X2CrNi19-11 austenitic stainless steels and cast parts as per RM 341-1 and RM 341-2					Cast parts As per NF EN 10213 and NF A 32-060				
	$R_{p0,2min}$ (MPa)	$R_{p0,2moy}$ (MPa)	$R_{mmin}$ (MPa)	$S_m$ (MPa)	$S$ (MPa)	$R_{p0,2min}$ (MPa)	$R_{p0,2ave}$ (MPa)	$R_{mmin}$ (MPa)	$S_m$ (MPa)	$S$ (MPa)
20	180	225	460	120	115	185	231	440	123	110
100	145	181	410	120	114	140	175	392	123	109
150	130	163	380	117	106					
200	118	148	360	106	100	105	131	344	95	95
250	108	135	350	97	97	95	119	335	86	86
300	100	125	340	90	90					
350	94	118	340	85	85					
400	89	111	337	80	80					
450	85	106	331	76,5	76,5					
500	81	101		73	73					
550	80	100		72						

Table 10. Tabular data for stress-strain curves applicable for cyclic elastoplastic analysis of 304L / 1.4307 components. The information may also be used for 304 / 1.4301.

$\varepsilon$ $T$	0,00042	0,00046	0,00054	0,00066	0,002	0,003	0,005	0,01	0,05
20 °C	-	-	-	131 MPa	163 MPa	175 MPa	190 MPa	210 MPa	280 MPa
100 °C	-	-	105 MPa	-	135 MPa	144 MPa	155 MPa	171 MPa	242 MPa
200 °C	-	86 MPa	-	-	111 MPa	118 MPa	126 MPa	139 MPa	204 MPa
300 °C	73 MPa	-	-	-	95 MPa	100 MPa	110 MPa	119 MPa	180 MPa

## APPENDIX 3 STRESS-STRAIN CURVES FOR P265GH AND P295GH

Virgin monotonic stress-strain curves for P265GH are shown in Figure 22 below and for P295GH in Figure 23 below. The information is collected from Appendix 3 in [2]. The information below is only given as a background for discussion of the experimental results presented within this report. It is not intended for any other use in relation to ratchet simulation.

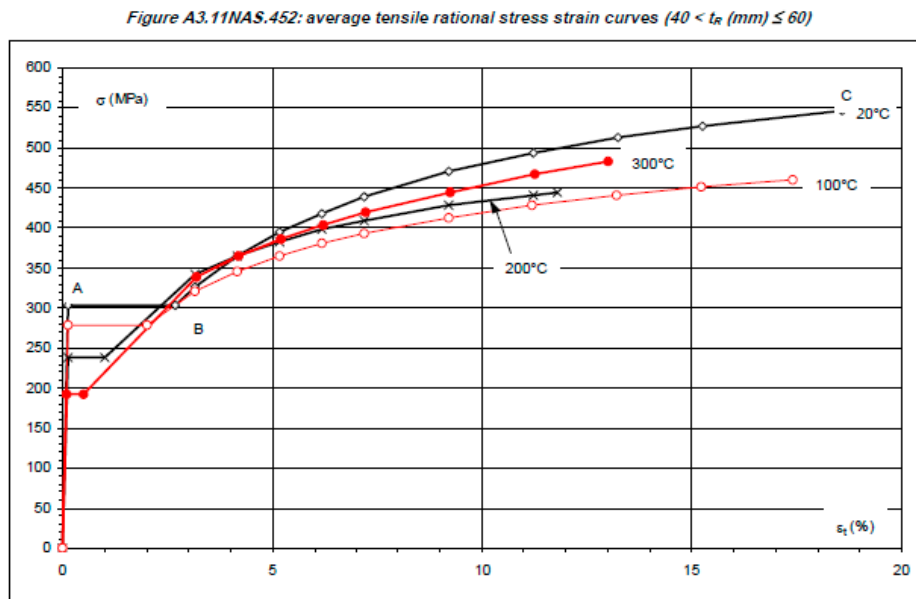


Figure 22. Stress-strain curves at room temperature and elevated temperatures for P265GH, from [2].

Figure A3.12NAS.452: average tensile rational stress strain curves ( $40 < t_R \text{ (mm)} \leq 60$ )

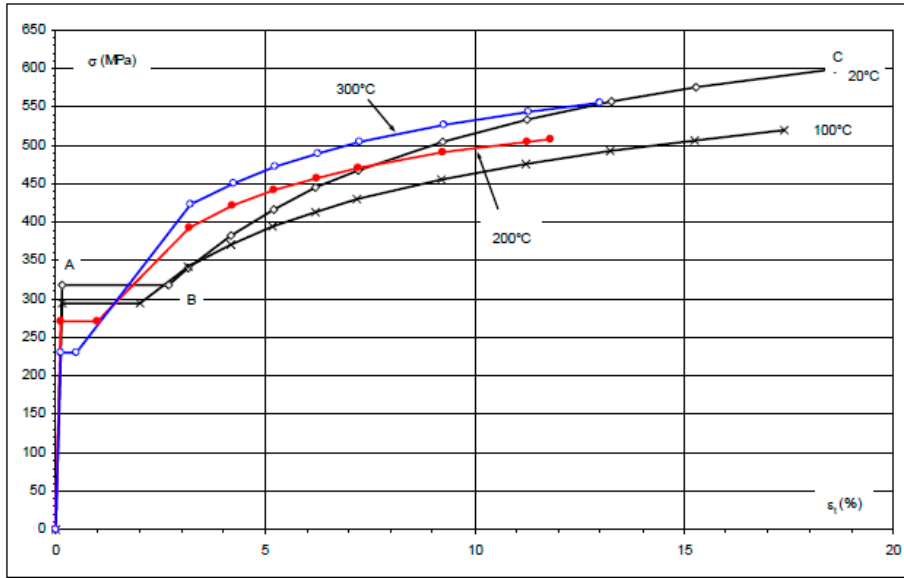


Figure 23. Stress-strain curves at room temperature and elevated temperatures for P295GH, from [2] .





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